United States of America

Department of Transportation — Federal Aviation Administration

Supplemental Type Certificate

Number SA00039DE

This certificate, issued to Oilamatic, Inc.

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part Regulations.

Original Product — Type Certificate Number: A7CE

Make: Cessna Aircraft Company
Model: 414, 414A

Description of Type Design Change:

Installation of Oilamatic Preoiler system in accordance with FAA approved Oilamatic, Inc. Drawing List No. 414400, No Revision, dated May 1, 1993, or later FAA approved revision.

Limitations and Conditions:

1. FAA approved Airplane Flight Manual Supplement, dated June 11, 1993, or later FAA approved revision is required.

2. Installation of this STC requires the installation of P/N C255 of STC SE00005DE.

Approval of this change in type design applies to the above model aircraft only.

(See Continuation Sheet, Page 3 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application May 7, 1993

Sale reissued :

Tale umended :

Date of issuance: June 11, 1993

By direction of the Administrator

RICHARD E. JENNINGS (Signature) Manager

Denver Aircraft Certification Field Office Northwest Mountain Region, Denver, Colorado

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

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Supplemental Type Certificate

(Continuation Sheet)

Number SA00039DE

This approval should not be extended to aircraft of this model on which other previously approved modifications are incorporated unless it is determined that the interrelationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of that aircraft.

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